

SUPPLEMENTAL TYPE CERTIFICATE

10074718

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

J & C AEROSPACE DESIGN AND CERTIFICATION SERVICES EEU Ltd.

31 BURNSIDE
MAGENTA CRESCENT
DUBLIN 9
D09 E866
IRELAND

EASA.21J.580

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.064

Type Certificate Holder: AIRBUS S.A.S.

Type: A318/A319/A320/A321

Model: A318-111/-112/-121/-122

A319-111/-112/-113/-114/-115

A319-131/-131/-133/-151N/-153N

A319-171N

A320-211/-212/-214/-215/-216

A320-231/-232/-233/-251N/-252N

A320-253N/-271N/-272N/-273N

A321-111/-112/-131/-211/-212

A321-213/-231/-232/-251N/-252N

A321-251NX/-252NX/-253N/-253NX

A321-271N/-271NX/-272N/-272NX

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 03 November 2020



Nicolas DUPREZ

Project Certification Manager – STC Coordinator



Description of Design Change:

SURVIVAL ELT INSTALLATION IN CABIN

Survival Emergency Locator Transmitter (ELT) Honeywell RESCU 406SG Series P/N 51090368-1 installation in cabin interior in enclosed overhead bin

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

Associated Technical Documentation:

Master Data List JCAERO/MDL/0638 issue 1,
or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -