

SUPPLEMENTAL TYPE CERTIFICATE

10072235

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

J & C AEROSPACE DESIGN AND CERTIFICATION SERVICES EEU Ltd.

31BURNSIDE MAGENTA CRESCENT DUBLIN 9 IRELAND

EASA.21J.580

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.064
Type Certificate Holder: AIRBUS S.A.S.

Type: A318/A319/A320/A321

Model: A320-214

Description of Design Change:

Re-activation of ISPS and associated seat change.

This Change is to re-activate and adapt the In-Seat Power Supply (ISPS) System on Airbus A320-214 aircraft post modification of currently installed B/E Aerospace Pinnacle 1019010 Series passenger seats in accordance with OEM Service Bulletin (SB) Collins 25-11000095-01.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 16 January 2020

Ludovic ARON
Head of Department
Large Aeroplanes





EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment: CS 25.831, 25.853, 25.869, 25.1431, 25.1529 at Amendment 22.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Master Data List: JCAERO/MDL/0403 Issue 1.3, or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

